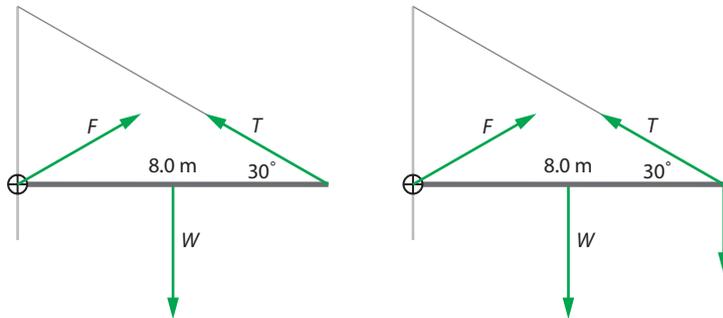


# Answers to exam-style questions

## Option B

1 ✓ = 1 mark

1 a i The forces are as shown in the left diagram:



The perpendicular distance between the axis and the line of the tension force is  $L \sin 30^\circ$ . ✓  
Rotational equilibrium by taking torques about an axis through the point of support gives:

$$W \times \frac{L}{2} = TL \sin 30^\circ \quad \checkmark$$

Hence  $W = T = 15 \text{ kN}$ . ✓

ii Translational equilibrium gives:  $T \cos 30^\circ = F_x$  and  $T \sin 30^\circ + F_y = 15 \text{ kN}$ . ✓

Hence  $T \cos 30^\circ = F_x = 12.99 \approx 13 \text{ kN}$  and  $F_y = 7.5 \text{ kN}$  so that the magnitude of  $F$  is

$$F = \sqrt{12.99^2 + 7.5^2} = 15 \text{ kN}. \quad \checkmark$$

And the direction to the horizontal is  $\theta = \tan^{-1} \frac{7.5}{12.99} = 30^\circ$ . ✓

b The critical case is when the worker stands all the way to the right. ✓

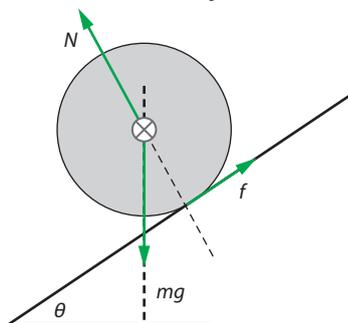
Rotational equilibrium in this case gives:  $W \times \frac{L}{2} + mgL = TL \sin 30^\circ$ . ✓

Solving for the tension gives:  $T = 16.7 \approx 17 \text{ kN}$ . ✓

2 a The forces are shown in the diagram and they are the weight of the cylinder,  $mg$ . ✓

The normal reaction,  $N$ . ✓

A frictional force  $f$ . ✓



b i Newton's second law for the translational motion down the plane is  $Mg \sin \theta - f = Ma$ . ✓

For the rotational motion by taking torques about the axis through the centre of mass is

$$fR = \left( \frac{1}{2} MR^2 \right) \alpha = \frac{1}{2} MRa \quad \checkmark$$

$$Mg \sin \theta - \frac{1}{2} Ma = Ma \quad \checkmark$$

From which the result follows.

$$\text{ii } f = Mg \sin \theta - Ma = 12 \times 9.8 \times \sin 30^\circ - 12 \times \frac{2}{3} \times 9.8 \times \sin 30^\circ = 19.6 \approx 20 \text{ N } \checkmark$$

c The rate of change of the angular momentum is the net torque.  $\checkmark$

$$\text{And this is } fR = 19.6 \times 0.20 = 3.92 \approx 4.0 \text{ Nm} . \checkmark$$

3 a i When the ring makes contact with the disc and while it is sliding, it exerts a frictional force on the disc but the disc exerts equal and opposite force on the ring.  $\checkmark$

Hence the net torque is zero and hence angular momentum is conserved.  $\checkmark$

$$\text{ii The initial angular momentum of the disc is } L = I\omega = \frac{1}{2}MR^2\omega = \frac{1}{2} \times 4.00 \times 0.300^2 \times 42.0 = 7.56 \text{ Js} . \checkmark$$

$$\text{After the ring lands the total angular momentum is } L = \frac{1}{2} \times 4.00 \times 0.300^2 \times \omega + 2.00 \times 0.300^2 \times \omega . \checkmark$$

$$\text{Hence } \frac{1}{2} \times 4.00 \times 0.300^2 \times \omega + 2.00 \times 0.300^2 \times \omega = 7.56 \text{ Js which gives } \omega = 21 \text{ rad s}^{-1} . \checkmark$$

$$\text{iii The initial kinetic energy is } E_K = \frac{1}{2}I\omega^2 = \frac{1}{2} \times \left( \frac{1}{2} \times 4.00 \times 0.300^2 \right) \times 42.0^2 = 158.76 \text{ J} . \checkmark$$

$$\text{The final is } E_K = \frac{1}{2} \times \left( \frac{1}{2} \times 4.00 \times 0.300^2 \right) \times 21.0^2 + \frac{1}{2} \times (2.00 \times 0.300^2) \times 21.0^2 = 79.38 \text{ J leading to a loss of } 79.38 \approx 79.4 \text{ J} . \checkmark$$

$$\text{b i } \alpha = \frac{\Delta \omega}{\Delta t} = \frac{21.0}{3.00} = 7.00 \text{ rad s}^{-2} \checkmark$$

$$\text{ii } \theta = \frac{1}{2} \alpha t^2 = \frac{1}{2} \times 7.00 \times 3.00^2 = 31.5 \text{ rad } \checkmark$$

$$\text{Which is } \frac{31.5}{2\pi} = 5.0 \text{ revolutions} . \checkmark$$

$$\text{iii } \Gamma = \frac{\Delta L}{\Delta t} \checkmark$$

$$\Gamma = \frac{2.00 \times 0.300^2 \times 21.0}{3.00} = 1.26 \text{ Nm } \checkmark$$

iv It is equal and opposite to that on the ring.  $\checkmark$

Because the force on the ring is equal and opposite to that on the disc.  $\checkmark$

$$\text{c The change in the kinetic energy of the ring is } \frac{1}{2} \times (2.00 \times 0.300^2) \times 21.0^2 = 39.69 \text{ J} . \checkmark$$

$$\text{And so the power developed is } \frac{39.69}{3.00} = 13.2 \text{ W} . \checkmark$$

$$\text{(This can also be done through } P = \Gamma \bar{\omega} = 1.26 \times \frac{21.0}{2} = 13.2 \text{ W} .)$$

4 a i The temperature at B doubled at constant volume so the pressure also doubles at  $p_B = 4.00 \times 10^5 \text{ Pa} . \checkmark$

$$\text{ii From } pV^{\frac{5}{3}} = c \text{ and } pV = nRT \text{ we find } p^{-\frac{2}{3}}T^{\frac{5}{3}} = c' . \checkmark$$

$$\text{Hence } (4.00 \times 10^5)^{-\frac{2}{3}} \times (600)^{\frac{5}{3}} = (2.00 \times 10^5)^{-\frac{2}{3}} \times T_C^{\frac{5}{3}} \text{ leading to } T_C = 455 \text{ K} . \checkmark$$

$$\text{iii The volume at B is } V_B = \frac{nRT_B}{p_B} = \frac{1.00 \times 8.31 \times 300}{2.00 \times 10^5} = 1.246 \times 10^{-2} \approx 1.25 \times 10^{-2} \text{ m}^3 . \checkmark$$

$$\text{And so } \frac{V_B}{T_B} = \frac{V_C}{T_C} \Rightarrow V_C = V_B \frac{T_C}{T_B} = 1.246 \times 10^{-2} \times \frac{455}{300} = 1.890 \times 10^{-2} \approx 1.89 \times 10^{-2} \text{ m}^3 . \checkmark$$

**b i**  $\Delta U_{AB} = \frac{3}{2} Rn\Delta T = +\frac{3}{2} \times 8.31 \times 1.00 \times 300 \checkmark$

$$\Delta U_{AB} = +3739 \approx +3.74 \times 10^3 \text{ J} \checkmark$$

**ii** This happens from C to A:  $W = -p\Delta V = 2.00 \times 10^5 \times (1.25 - 1.89) \times 10^{-2} = -1280 \text{ J}$  and the change in internal energy is  $\Delta U_{AB} = \frac{3}{2} Rn\Delta T = \frac{3}{2} \times 8.31 \times 1.00 \times (300 - 455) = -1932 \text{ J} \checkmark$

$$\text{Hence } Q = \Delta U + W = -1932 - 1280 = -3212 \approx -3.21 \times 10^3 \text{ J} \checkmark$$

**c** Any heat engine working in a cycle cannot transform all the heat into mechanical work.  $\checkmark$

And this engine rejects heat into the surroundings as it should.  $\checkmark$

**5 a i** A curve along which no heat is exchanged.  $\checkmark$

**ii** An adiabatic expansion involves a piston moving outwards fast.  $\checkmark$

Hence molecules bounce back from the piston with a reduced speed and hence lower temperature.  $\checkmark$

**b** The product pressure  $\times$  volume is constant for an isothermal.  $\checkmark$

This is the case for points A and C (product is 100 J).  $\checkmark$

And the same is true for any other point on the curve, for example at  $p = 2.00 \times 10^5 \text{ Pa}$ ,  $V = 0.50 \times 10^{-3} \text{ m}^3$ .  $\checkmark$

**c i**  $\frac{V_A}{T_A} = \frac{V_B}{T_B} \Rightarrow T_B = T_A \frac{V_B}{V_A} \checkmark$

$$T_B = 300 \times \frac{0.38}{0.20} = 570 \text{ K} \checkmark$$

At C  $T_C = 300 \text{ K}$  since AC is isothermal.  $\checkmark$

**ii** Using data at A:  $n = \frac{pV}{RT} = \frac{5.00 \times 10^5 \times 0.20 \times 10^{-3}}{8.31 \times 300} \checkmark$

$$n = 4.01 \times 10^{-2} \checkmark$$

**d i** Energy is transferred out of the gas along C to A.  $\checkmark$

From  $Q = \Delta U + W$  and  $\Delta U = 0$  we find  $Q = -160 \text{ J}$ .  $\checkmark$

**ii** This happens from A to B:  $W = 5.00 \times 10^5 \times (0.38 - 0.20) \times 10^{-3} = 90 \text{ J}$  and

$$\Delta U = \frac{3}{2} \times 8.31 \times 4.01 \times 10^{-2} \times (570 - 300) = 135 \text{ J} \checkmark$$

And so  $Q = 135 + 90 = 225 \text{ J}$ .  $\checkmark$

**iii**  $W_{BC} = -\Delta U_{BC}$  (since BC is an adiabatic).  $\checkmark$

And  $\Delta U_{BC} = -\Delta U_{AB} = -135$  (since AC is an isothermal).  $\checkmark$

OR

Since for the whole cycle  $\Delta U = 0$ , the net work is  $Q_{\text{in}} - Q_{\text{out}} = 225 - 160 = 65 \text{ J}$ .  $\checkmark$

And  $W_{\text{net}} = W_{AB} + W_{BC} + W_{CA} \Rightarrow 128 = 90 + W_{BC} - 160 \Rightarrow W_{BC} = 198 \text{ J}$ .  $\checkmark$

**iv** The efficiency is  $e = \frac{W_{\text{net}}}{Q_{\text{in}}} = \frac{65}{225} = 0.290$ .  $\checkmark$

**6 a** For an adiabatic,  $pV^{\frac{5}{3}} = c$  and since  $V = \frac{nRT}{p}$ .  $\checkmark$

$$\text{we find } p \left( \frac{nRT}{p} \right)^{\frac{5}{3}} = c \checkmark$$

Raising to the 3<sup>rd</sup> power gives  $p^3 \left( \frac{nRT}{p} \right)^5 = c^3$  and so the result.  $\checkmark$

**b i** From  $\frac{T^5}{p^2} = \text{constant}$  we find  $\frac{320^5}{(2.0 \times 10^5)^2} = \frac{T^5}{(2.0 \times 10^6)^2}$ .  $\checkmark$

$$T = 320 \times \left( \frac{2.0 \times 10^6}{2.0 \times 10^5} \right)^{\frac{2}{3}} = 803.8 \approx 800 \text{ K} \checkmark$$

ii From  $pV^{\frac{2}{3}} = \text{constant}$  we find.  $\checkmark$

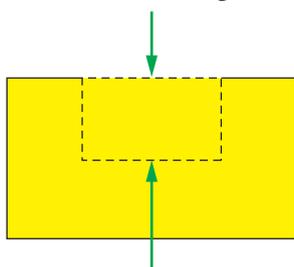
$$V = \left( \frac{2.0 \times 10^5}{2.0 \times 10^6} \right)^{\frac{3}{2}} \times 0.40 = 0.10 \text{ m}^3 \checkmark$$

c The number of moles is  $n = \frac{pV}{RT} = \frac{2.0 \times 10^5 \times 0.40}{8.31 \times 320} = 30. \checkmark$

$$\Delta U = \frac{3}{2} Rn\Delta T = \frac{3}{2} \times 8.31 \times 30 \times (804 - 320) \checkmark$$

$$\Delta U = 0.181 \text{ MJ} \checkmark$$

7 Delineate a rectangular region in the liquid whose top surface is the free surface of the liquid and has area  $A$ .  $\checkmark$



Equilibrium demands that weight = net upward force.  $\checkmark$

In other words that  $\rho Ahg = pA - p_0A$ .  $\checkmark$

From which the result follows.

b i In free fall gravity “disappears” and so the pressure is just the atmospheric pressure.  $\checkmark$

ii When the liquid accelerates upwards there is an additional force pushing the liquid upwards and so the pressure increases.  $\checkmark$

c A body immersed in a fluid experiences an upward force that is equal to the weight of the displaced liquid.  $\checkmark$

d i Equilibrium demands that  $\rho_{\text{wood}}Vg = \rho_{\text{water}} \times 0.75Vg$ .  $\checkmark$

$$\text{Hence } \rho_{\text{wood}} = \rho_{\text{water}} \times 0.75 = 750 \text{ kg m}^{-3} \checkmark$$

ii Equilibrium demands that  $\rho_{\text{wood}}Vg = \rho_{\text{oil}} \times 0.82Vg$ .  $\checkmark$

$$\text{Hence } \rho_{\text{oil}} = \frac{\rho_{\text{wood}}}{0.82} = \frac{750}{0.82} = 914.6 \approx 910 \text{ kg m}^{-3} \checkmark$$

8 a i  $p_0 + \rho gz = p_0 + \frac{1}{2} \rho v^2$  hence  $v = \sqrt{2gz}$   $\checkmark$

$$v = \sqrt{2 \times 9.8 \times (220 + 40)} = 71.4 \approx 71 \text{ m s}^{-1} \checkmark$$

ii That the flow is laminar,  $\checkmark$

and there are no losses of energy.  $\checkmark$

b The flow rate is given by  $Q = Av = \pi R^2 v$ .  $\checkmark$

$$\text{Hence } Q = \pi \times (0.25)^2 \times 71.4 = 14 \text{ m}^3 \text{ s}^{-1} \checkmark$$

c i  $p = p_0 + \rho gh = 1.0 \times 10^5 + 1000 \times 9.8 \times 40$   $\checkmark$

$$p = 4.9 \times 10^5 \text{ Pa} \checkmark$$

ii The pressure is given by  $p_0 + \rho gz = p + \frac{1}{2} \rho v^2$  where the speed can be found from the flow rate (i.e. the continuity equation)  $\pi \times (0.65)^2 \times v = 14.02 \approx 14 \text{ m}^3 \text{ s}^{-1}$ .  $\checkmark$

i.e.  $v = 10.56 \approx 11 \text{ ms}^{-1}$ . ✓

And hence

$$p = p_0 + \rho gz - \frac{1}{2} \rho v^2 = 1.0 \times 10^5 + 1000 \times 9.8 \times 40 - \frac{1}{2} \times 1000 \times 10.56^2 = 4.36 \times 10^5 \approx 4.4 \times 10^5 \text{ Pa} \quad \checkmark$$

**d** The speed at depth  $h$  is  $v = \sqrt{2gh}$ . ✓

The flow rate is  $Q = Av = \pi R^2 \sqrt{2gh}$  and has to equal  $0.40 \text{ m}^3 \text{ s}^{-1}$ . ✓

$$\text{Hence } h = \frac{1}{2g} \left( \frac{0.40}{\pi R^2} \right)^2 = \frac{1}{2 \times 9.8} \left( \frac{0.40}{\pi \times 0.03^2} \right)^2 = 7.2 \text{ m}. \quad \checkmark$$

**9 a** The left side is connected to the holes in the tube past which the air moves fast. ✓

Hence the pressure there is low and the liquid is higher. ✓

**b** Call the pressure at the top of the left column  $p_L$  and that on the right  $p_R$ . Then

$$p_L + \rho_{\text{air}} gz + \frac{1}{2} \rho v_L^2 = p_R + \rho_{\text{air}} gz + \frac{1}{2} \rho v_R^2 \quad \text{and with } v_L = 0; v_R = v, \quad \checkmark$$

$$\text{it becomes } v = \sqrt{\frac{2(p_L - p_R)}{\rho_{\text{air}}}}. \quad \checkmark$$

But  $p_L - p_R = \rho gh$  which gives the result. ✓

**c** 
$$v = \sqrt{\frac{2\rho gh}{\rho_{\text{air}}}} = \sqrt{\frac{2 \times 920 \times 9.8 \times 0.25}{1.20}}. \quad \checkmark$$

$$v = 61.3 \approx 61 \text{ m s}^{-1}. \quad \checkmark$$

**10 a** Smooth streamlines. ✓

Closer together above the aerofoil. ✓

**b i** From  $p_L + \rho gz + \frac{1}{2} \rho v_L^2 = p_U + \rho gz + \frac{1}{2} \rho v_U^2$  we obtain  $\Delta p = p_L - p_U = \frac{1}{2} \rho v_U^2 - \frac{1}{2} \rho v_L^2$ . ✓

$$\text{Hence } F = A \Delta p = A \left( \frac{1}{2} \rho v_U^2 - \frac{1}{2} \rho v_L^2 \right) = 8.0 \times \frac{1}{2} \times 1.20 \times (85^2 - 58^2) = 18.53 \approx 19 \text{ kN}. \quad \checkmark$$

**ii** That the area above and below the foil are equal/that the flow is laminar. ✓

**c** The net upward force on the foil is about 16 kN and this is an estimate of the downward force on the fuselage. ✓  
Ignoring effects of torque. ✓

**d i** The streamlines are no longer smooth but become eddy like and chaotic. ✓

**ii** Everywhere on the top side of the aerofoil and especially to the right. ✓

**iii** It will be drastically reduced. ✓

**11 a** In undamped oscillations the energy is constant and so the amplitude stays the same. ✓

In damped oscillations energy is dissipated and the amplitude keeps getting smaller. ✓

**b i**  $8.0 \text{ s}$  ✓

**ii** Correct readings of amplitudes. ✓

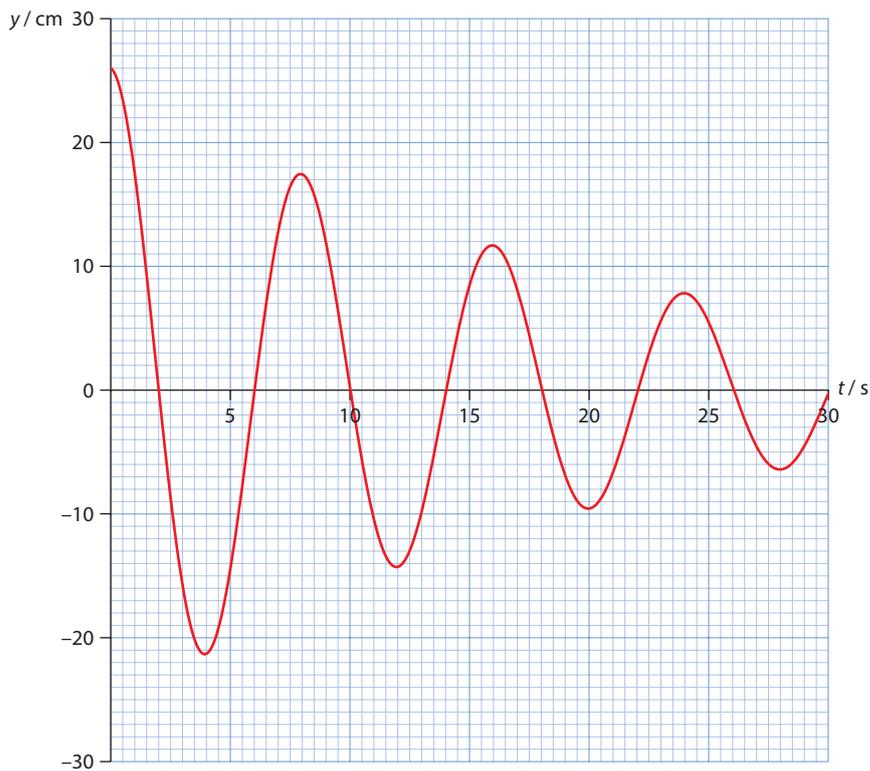
$$Q = 2\pi \frac{26^2}{26^2 - 22^2} \quad \checkmark$$

$$Q \approx 22 \quad \checkmark$$

**c i** Amplitude reducing more every cycle. ✓

Period staying essentially unchanged/very slightly increases. ✓

**ii** It will decrease. ✓



**d**  $Q = 2\pi \frac{5.0}{5.0 - 4.6} \checkmark$

$Q \approx 79 \checkmark$

**12 a** All oscillating systems have their own natural frequency of oscillation.  $\checkmark$

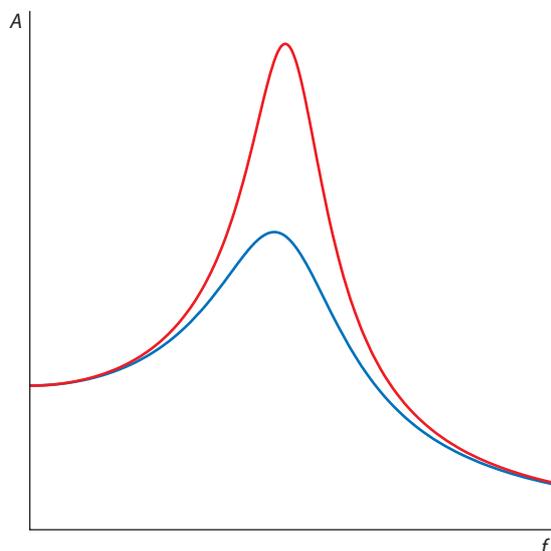
When a periodic external force is applied to the system the amplitude of oscillation will depend on the relation of the external frequency to the natural frequency.  $\checkmark$

The amplitude will be large when the frequency of the external force is the same as the natural frequency in which case we have resonance external.  $\checkmark$

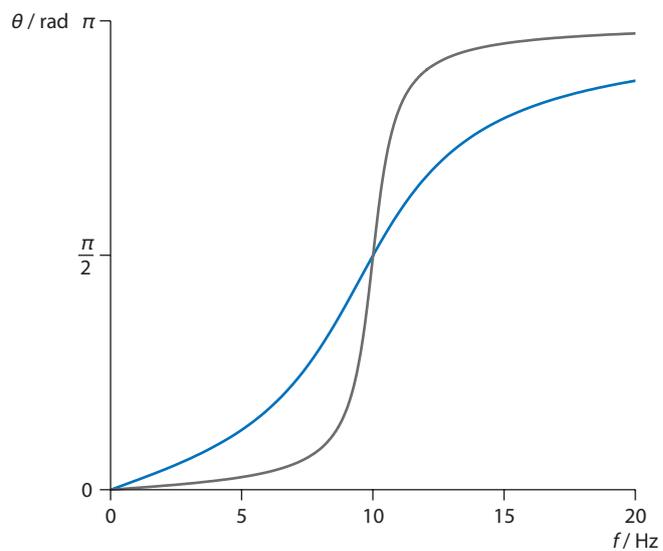
**b** Wider and lower curve.  $\checkmark$

With peak shifted slightly to the right.  $\checkmark$

See curve in blue.



- c i** Same intersection point. ✓  
Less steep. ✓  
See curve in blue.



- ii** 10 Hz ✓